

National Aeronautics and Space Administration

UAS in the NAS Flight Test Series 4 Overview

UAS Integration in the NAS Project

Jim Murphy: Project Engineer, Integrated Test and Evaluation

12 May 2016



Flight Test Series 4 Goals

 Conduct flight tests in a relevant environment to contribute to the validation of the final Phase 1 DAA MOPS

 Develop and maintain the infrastructure for a relevant test environment for UAS research



Flight Test Series 4 Research Objectives

- Validate DAA requirements in stressing cases that drive MOPS requirements, including: High-speed cooperative intruder, Low-speed non-cooperative intruder, high vertical closure rate encounter, and Mode C only intruder (i.e., without ADS-B)
- Validate TCAS/DAA alerting and guidance interoperability concept in the presence of realistic sensor, tracking and navigational errors and in multiple-intruder encounters against both cooperative and non-cooperative intruders
- Validate 'Well Clear Recovery' guidance in the presence of realistic sensor, tracking and navigational errors

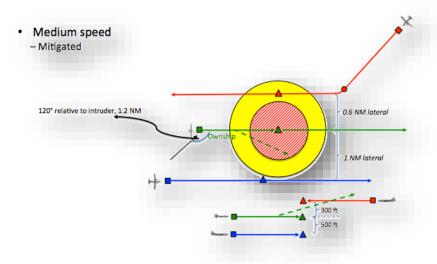


Flight Test Series 4 Test Environment

- Support validation of final phase 1
 DAA and Radar MOPS
 - Sensor noise, uncertainty
 - Navigation system errors, state data uncertainty,
 - Wind compensation
 - Support SC-228 V&V activities
- Refine DAA alerting and maneuver guidance algorithms
 - Stressing encounters
 - More complex multi-intruder encounters
 - TCAS/DAA interoperability
 - Well clear recovery
 - Mixed intruder equipage

DAA Scripted Encounters

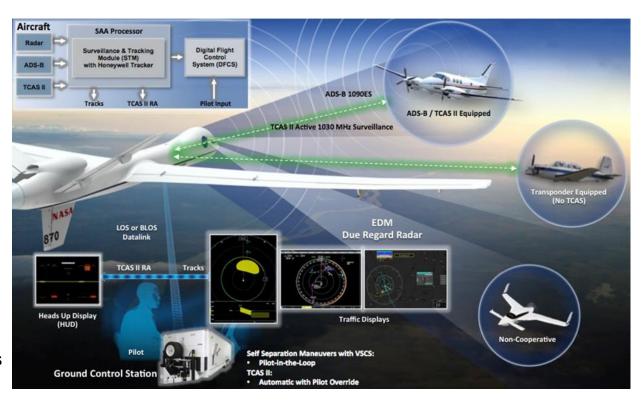
- Live Ownship with Coop and Non-Coop Sensors
- Live Intruder(s)





FT4 Test System Concept

- FT4 leverages off of FT3 & ACAS Xu
- DAA Scripted Encounters only
- Data Collection for SUT:
 - JADEM
 - o DAIDALUS
 - o CPDS
 - o TCAS
 - Radar
 - Honeywell
 - o SC-228
- Sensor Evaluation
 - O ADS-B
 - TCAS Hybrid Surveillance
 - o EDM Radar
 - o Mode C
 - Non-cooperative Target
- Flight Test Envelope Increased
 - Low Speed
 - o High Speed
- Increased Intruder Requirements
 - o GIII
 - o TG-14
 - o C-12
- Multiship Encounters 1v4
- R-2515 Test Area similar to FT3



271 encounters planned for FT4 – 20% increase from FT3

Each encounter takes approximately 10 min and planning to complete ~20 encounters per flight day



Integration Roles & Responsibilities Summary

NASA – AFRC (UAS-NAS / IT&E)

- Provide LVC-DE Infrastructure
- Provide Intruder Aircraft (T-34, G-III, TG-14, King Air)
- Provide Ownship Aircraft (Ikhana)
- Test Conductor Station (SAF)

NASA - ARC (UAS-NAS / IT&E)

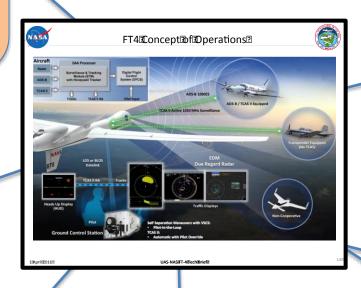
- Provide HLA infrastructure
- RUMS Server
- Video Distribution to LaRC

SC-228

Devise Encounter matrix

NASA

Non-NASA



NASA - ARC (UAS-NAS / SSI)

- Provide JADEM DAA
- Provide Uncertainty model
- Devise Encounter matrix

NASA - LaRC (UAS-NAS / SSI)

- Provide DAIDALUS DAA
- Devise Encounter matrix

Honeywell

- Provide instrumented TCAS
 Il equipped intruder aircraft
- Fusion Tracker
- Post Flight Processing
- Devise Encounter matrix

GA-ASI

- Provide proof of concept DAA system (EDM DRR, SAAP, etc.)
- CPDS Display and IO Server
- Devise Encounter matrix

Flight Test 4 Milestones/Key Activities **▲ SA T**imeline Not To Scale M (11/13) ERT Decision to delete FT4 Full **Development Activities** Mission FT4 Requirements TIM Scripted Encounters Flight Block (F2F Peer Review) **Intruder Aircraft Mods** FT4 Design TIM Review / Tech Brief (F2F Peer Review) NASA808 ADS-B Modifications 4/11 2/5 **NASA808 DGPS Modifications** NASA801 DGPS Modifications **NASA856 ADS-B Modifications** 4/11 4/21 **Ikhana FT4 Modifications** 12/7 Ground Integration Tests/V&V 2/8 2/25 Integrated LVC/Ikhana Ground Testing (CST) 2/29 ARC & AFRC JADEM Regression Testing 3/18 Flight Test Plan Complete 4/4 4/15 2/19 **Tech Brief** System Checkout Flights / Env. Expansion Flights 4/1 **Support Aircraft Airworthiness / Tech Brief Scripted Encounters Flight Test** 6/30 Review 4/19



FT4 Participating Aircraft

Aircraft Required Systems

- o ADS-B Out
- o Mode C or S Transponder
- o GPS
- o DGPS w/ Recording
- VHF Comm Radio (x2)

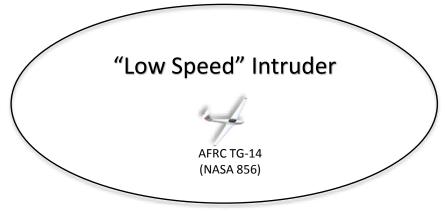
s	Aircraft	Role	EDM DRR	ADS-B	DGPS	TCAS-II	TCAS-I	Mode S	Mode C
r		Ownship NASA AFRC Ikhana UAS	✓	✓	✓	\		✓	✓
		Primary Intruder Honeywell Beechcraft King Air C90		✓	✓	✓		✓	
		Secondary Intruder NASA AFRC T-34		✓	✓		>	/	
		Secondary Intruder NASA AFRC King Air		✓	/		>	✓	
NEW		High Speed / Secondary Intruder NASA AFRC GIII		✓	✓	✓		✓	
NEW		Low Speed / Secondary Intruder NASA AFRC TG-14		✓	✓			✓	✓
NEW		Secondary Intruder USAF C-12 (Mode C Only)			✓				✓

Ownship – Ikhana only

New Intruders – Low Speed/RCS, High Speed, Mode C only

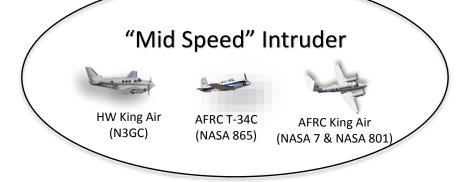


Aircraft Performance



TG-14

- 100 KGS
- Small RCS (radar specific)
- Low altitude flights (5000ft-10000ft)



T-34

- Up to 250 KGS
- Medium RCS (radar specific)
 King Airs
- Up to ~280 KGS
- Large RCS (radar specific)

"High Speed" Intruder

AFRC G-III
(NASA 808)

G-III

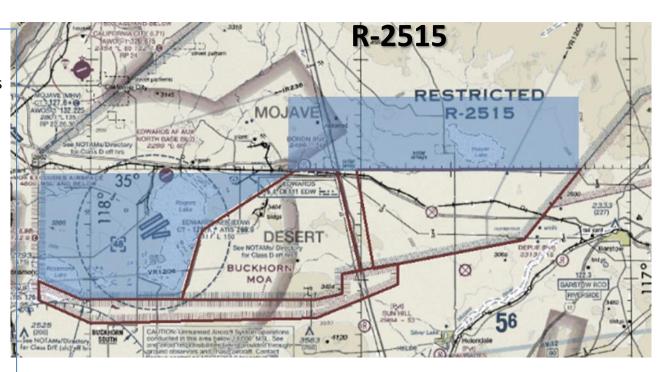
Up to ~500 KTAS at 20k ft

NASA

FT4 Test Area

Airspace Planning

- Primarily Mercury Spin, 4 Corners
 & Buckhorn MOA (red outline)
- 1,000 ft AGL (4.2K ft) to 20K ft MSL
- Extensions (west / north) may be requested real time for encounters that need the additional airspace
- Ops outside of test area (blue shaded areas) are planned to be performed early (before 0800) when airspace is relatively empty
- Operations between 0600 and 0700 are under Joshua control and have less geographical constraints



Airspace Extensions (Blue Shaded Areas)

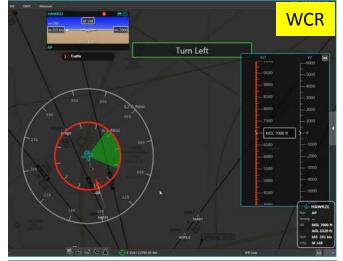
- Conducted early 0600-0800 preferably
- Pre-coordinated 24-48 hours in advance
- Requested real-time with SPORT (after 0700)

Ikhana must remain within R-2515 at all times. Intruder aircraft can use Buckhorn MOA, plus areas shaded in blue.



FT4: NASA DAA Suite: JADEM/VSCS

- Java Architecture for DAA Extensibility and Modeling (JADEM) software:
 - Subscribes to LVC GW for ownship and intruder flight state data (flight state message).
 - Subscribes to LVC GW for TCAS RA messages.
 - Contains DAA algorithm for conflict detection and resolution.
 - Contains DAA/TCAS interoperability logic.
 - Publishes DAA conflict resolution as heading bands and altitude tape keep out zones.
 - Publishes Well Clear Recovery (WCR) as suggestive guidance.
- Vigilant Spirit Control Station (VSCS) software:
 - Developed by AFRL as an integrated ground control station for command and control of UAS.
 - VSCS Tactical Situation Display (TSD) augmented by AFRL with DAA display elements to support NASA Human Systems Integration (HSI) objectives.
 - VSCS TSD serves as the Cockpit Display for Traffic Information (CDTI) for SSI-West experiment.
 - VSCS TSD displays DAA and WCR guidance produced by JADEM.
 - VSCS TSD displays TCAS RA aural and visual alerts.







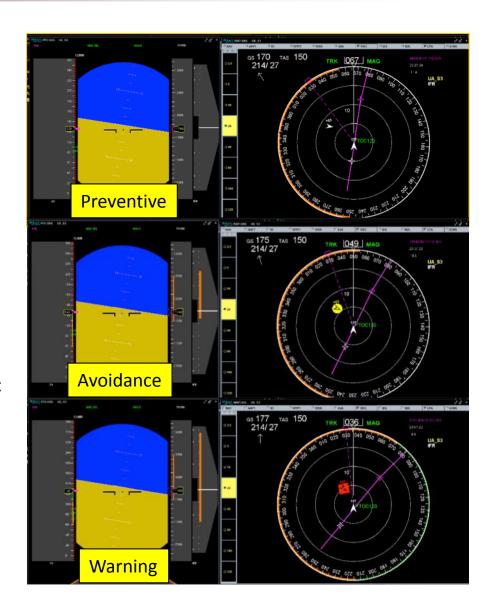
FT4: NASA DAA Suite: DAIDALUS/MACS

DAIDALUS:

- Subscribes to LVC GW for ownship and intruder flight state data (flight state message).
- Subscribes to LVC GW for TCAS RA messages.
- Contains DAA algorithm for conflict detection and resolution.
- Contains DAA/TCAS interoperability logic.
- Provides DAA avoidance and recovery bands.
- Provides WCR band.

MACS:

- Adapted for use as a DAA test bed.
- Displays DAA guidance from DAIDALUS on generic HSI, VSI, and altitude instruments.
- VSI DAA bands replaced by TCAS guidance during RAs.
- Displays WCR band.

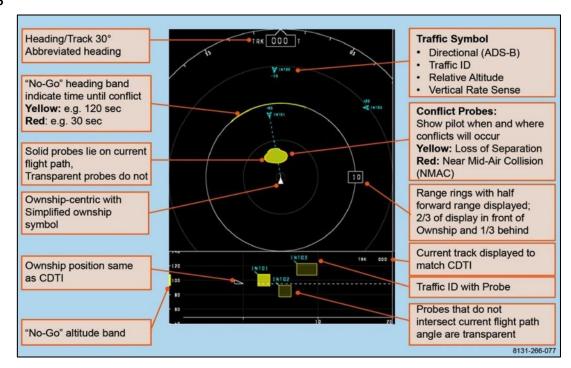


NASA

FT4: GA-ASI CPDS

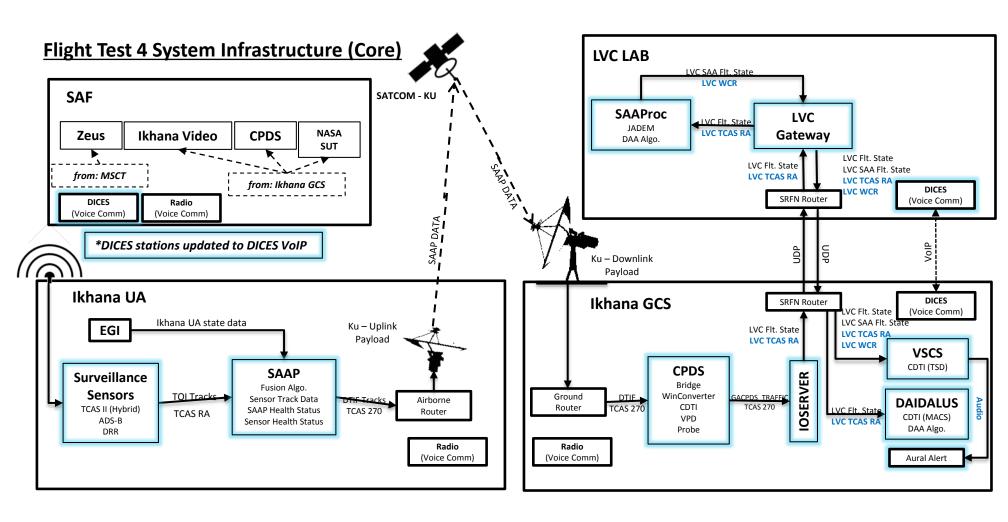
CPDS:

- Receives ownship and surveillance tracks from the SAAP.
- Displays Ownship and Surveillance track data on the CDTI display and VPD.
- Processes surveillance tracks through its DAA algorithm to create conflict probes displayed on CDTI and VPD.
- Provides engineering display (Winconverter) with additional data on ownship and intruder.
- Bridge module provides option to select adding UTC offset to the received A735B with readout to verify offset.
- Bridge module includes fractional seconds in the time put in to the GA_CPDS traffic topic. Allowing IOServer to forward time.
- Winconverter proxy panel to suppress certain data to CPDS domain.
- Winconverter capability to send DAA alerts to HUD (if connected to flight network).
- CDTI added North-up option.





FT4: Core System Infrastructure





Encounter Planning Process

Researcher Evaluation

Using local constraints (E.G. Lat/Long)
Simulate/Evaluate

- Mission success / original intent
- Expected results

JADEM

DAIDALUS

CPDS

TCAS

Radar

Honeywell

SC-228

Initial Requirements

OWG & SSWG

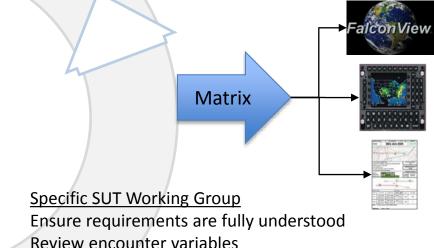
Safety compliance

- Specified maneuvers
- Required safety constraints
- Update hazards

Mission Success

- Test Objective Table
- Training material
- Simulation Video (as applicable)

Measurable and Achievable



- Test objective Tables

- Flight Cards

Ensure compliance with constraints and limitations

۳,		
	Encounter Set	All encounters with same variables
	Configuration	Geometries / Equipment
	Test Objectives	Primary TOs
	Success Criteria	What determines success in flight
	Test Methods	Procedures to achieve success criteria
ة	Evaluation Criteria	Post flight analysis / data collection

Derived from AFTC Test Planning Guide



Encounter Tolerance Development

ACAS Xu

- Probalistic approach 99.7%
- Min offsets:
 - 200' vertical
 - 0.5 nmi lateral
 - ±5, 8, 10 sec
- Altimeter & navigation calibration
- Mission Rules



 Assuming a separation threshold, an integration over PDFs (letting δ be small) yields the desired probability calculation:



- Results:
 - Probability(<100 Feet Separation | 200 Feet Planned) = 0.0689
 Probability(<20 Feet Separation | 200 Feet Planned) = 0.0032
 - Probability(<20 Feet Separation | 200 Feet Planned) = 0.0032
 Probability(<100 Feet Separation | 200 Feet Planned) = 0.0016
- Probability(<20 Feet Separation | 300 Feet Planned) = 1.5e-05
- Probability(<100 Feet Separation | 500 Feet Planned) = 1.3e-01
- · Planned (known) Vertical Separation:
- 500ft and 200ft
- Normal Distribution based on Standard Deviation
- Standard Deviation of 43ft
- Flight test calibration legs consider errors between aircraft
 Allows for up to 25 feet altitude difference (each within 12.5 feet of "true")
- Three standard deviations contains 99.7% of the probability
 We can safely assume 10 feet SD for instrument error
- GA Autopilot Altitude Mode standard deviation
- Request for Information
 Current Assumptions on SD to perform analysis
- 60 Feet SD for manned aircraft

ACAS – FT3

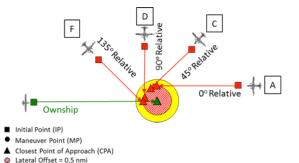
- High Confidence with ACAS-Xu offsets
- Mission Success criteria closely approximated ACAS-Xu
- Nav/Cal- not measureable, no added value or increase safety margins

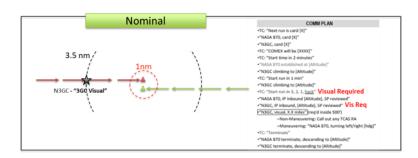
FT3

- Deterministic approach
 - Leveraged off of ACAS Xu
- Min offsets:
 - 200' vertical
 - 0.5 nmi lateral
 - ±5, 8, 10 sec
- · Altimeter calibration
- Mission Rules

SS Alerting Boundary=0.75 nmi

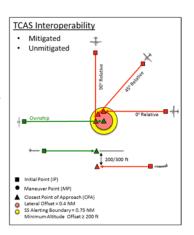
Minimum Altitude Offset ≥ 200 ft

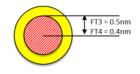




FT4

- Deterministic approach
 - Leveraged off of FT3
- Min offsets:
 - 200' vertical
 - 0.4 nmi lateral
 - ±5, 8, 10 sec (DAA Real Time)
- Altimeter calibration
- Mission Rules





Decreased to 0.4 nm horizontal separation. Required to achieve TCAS alerting below 10K

FT3 - FT4

- · High Confidence with previous test planning criteria
- Decrease from 0.5 to 0.4 nm horizontal offset for mission success and;
- Engineering & crew judgment min acceptable levels of safety are maintained

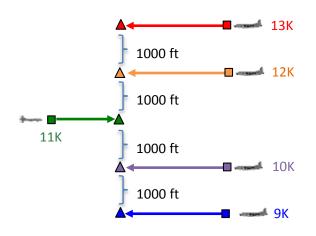
NASA

Ikhana FT-4 Mission Flights

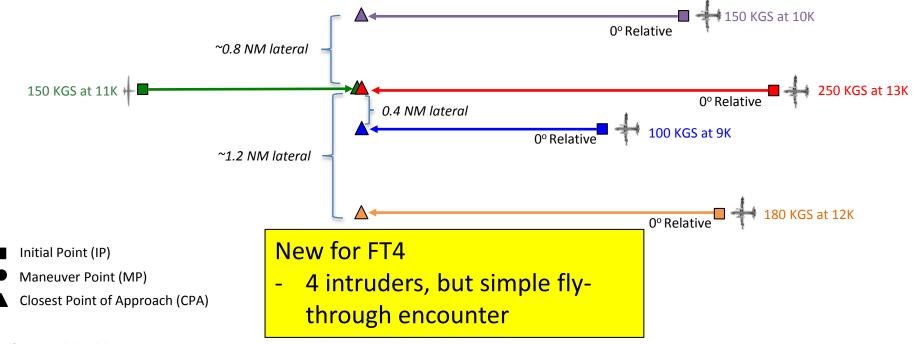
- 15 Flights Planned
 - Tuesdays and Thursdays, late April through June
- Up to 6.5 hour flights planned
- Single and Multiple Intruders
 - 1, 2, or 4 manned aircraft
 - Honeywell, NASA AFRC, and/or AF aircraft
- Objectives
 - Using various geometries, closing airspeeds, and altitudes
 - Demonstrate FT-4 system performance
 - Within approved FT-4 test envelope
 - Demonstrate various Self-Separation display systems
 - GA-ASI: Conflict Prediction and Display System (CPDS)
 - NASA: JADEM and DAIDALUS



Multiship (7) Four Intruders

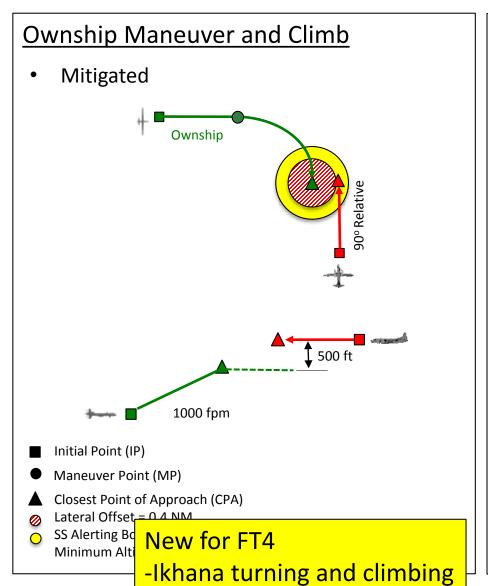


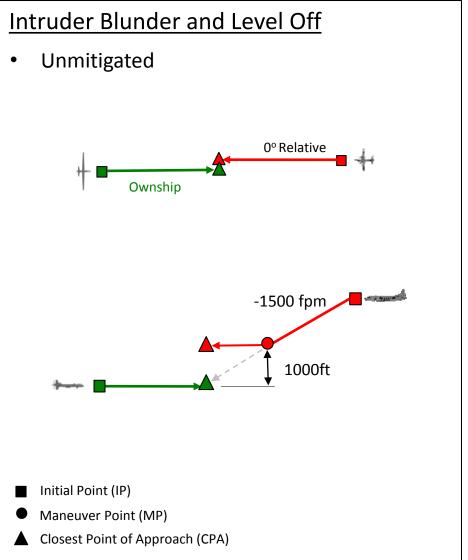
- Mixed speeds
 - Unmitigated





Medium Speed Intruder (4)



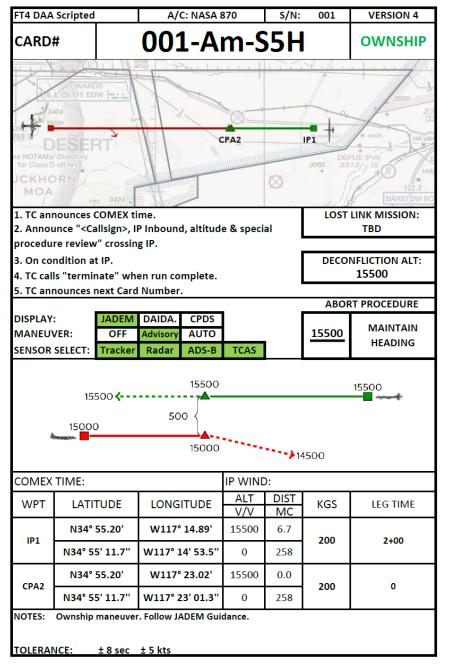


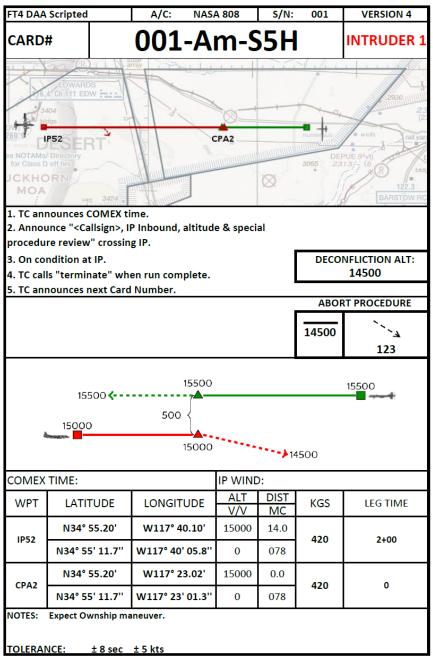
Same encounter variables as FT3

7. SC228 - #277-279, #280-282



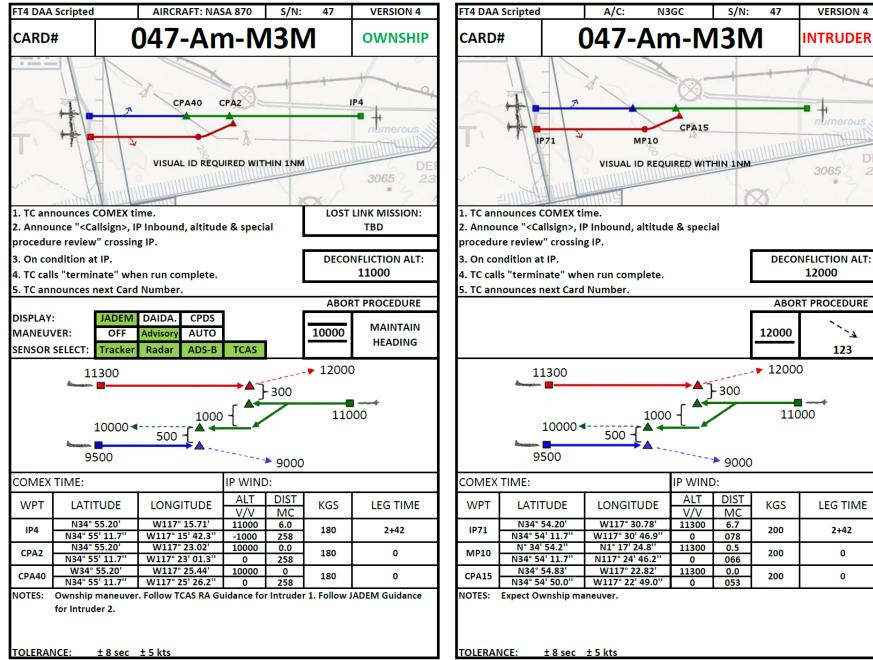
Single Intruder Test Card







Multiship Intruder Test Card (1)



VERSION 4

INTRUDER 1

12000

123

LEG TIME

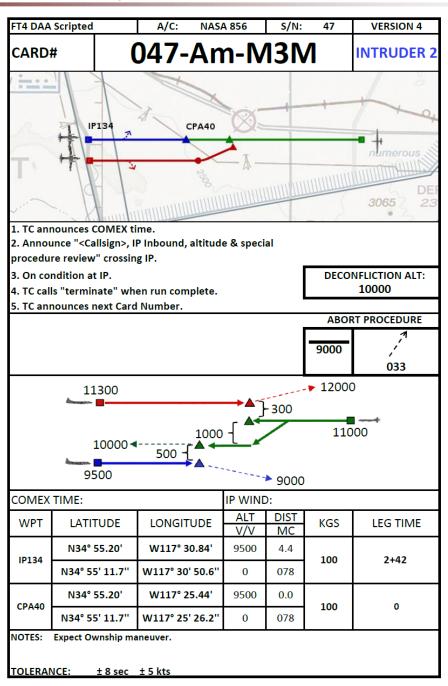
2+42

0

0



Multiship Intruder Test Card (2)





Minimum Separation

- The minimum geospatial offsets planned are 200 feet vertical and 0 foot horizontal (although not simultaneously during any test run)
- Test encounters with a minimum vertical separation of less than 500 feet will include a lateral offset of at least 2430 feet (0.4 nmi)
 - Allows for some built-in safety margin that still meets well-clear volume requirements and test data collection objectives
- Test encounters more than or equal to 500 feet vertical offset may have a 0 foot horizontal offset.
- All participating aircraft will ensure the aircraft altimeter system meets manufacturer calibration specifications and requirements for normal operation in the NAS.
- A maximum of 608 feet (0.1 nmi) navigation error (GPS derived position) is allowed for each aircraft based on the system's built-in navigation accuracy readout.



Ikhana FT-4 Nominal Encounter Mission Flow

- T-1: Day before detailed brief
 - Protect crew rest for back-to-back flights possible if back up dates are utilized
- T-0 All: All crew pre-flight delta brief
- T-0 Ikhana: Individual aircraft crew brief, as required
- Each aircraft take-off to arrive within R2515 at required time
 - 0600 ish for Ikhana
- Ikhana and manned aircraft perform altitude calibration, as required
- Prior to each encounter verify encounter, configuration, visual ID requirement, expected behavior, and abort procedures per UAS-NAS encounter cards and Ikhana flight cards



Ikhana FT-4 Nominal Encounter Mission Flow (cont)

- Execute Encounter
- Between Encounters
 - Maneuver Mode: Advisory/Off
 - Maintain previous encounter deconfliction altitude until laterally separated and as directed by TD for upcoming encounter
 - Configure for next encounter set up
- Execute additional encounters per flight cards
- RTB, as appropriate
- Land
- De-brief



FT4 Mission Management Responsibilities

Test Conductor

 Manages mission execution over voice communication on mission net (VHF radio) with Ikhana pilot, intruder pilots, and SPORT

Test Director

- Primary liaison with test conductor
- Manages voice communication on test team net with Ikhana mission director, SOR, LVC, local agencies, and non-local agencies (as required)

Mission Directors

 Communicates directly with co-located pilots within GCS, test director, and other agencies (as required)

LVC

 Communicates with test director on test team net to provide real-time discussions on live virtual constructive performance (as required)

Senior Operations Representative (SOR)

Independent representative of the Code O director with authority to stop a test if it is
deemed that the test team is making unauthorized changes to the briefed test cards, is
confronted with unanticipated events that should be addressed on the ground, or other
issues that might arise that warrant a test to be stopped and reevaluated in a less dynamic
environment than in the middle of a test mission

Range Control Officer

 RCO monitors and makes coordination calls with Armstrong COMM and the SAF DATR mission system ops folks (as required)



FT4 Mission Support Software



Zeus

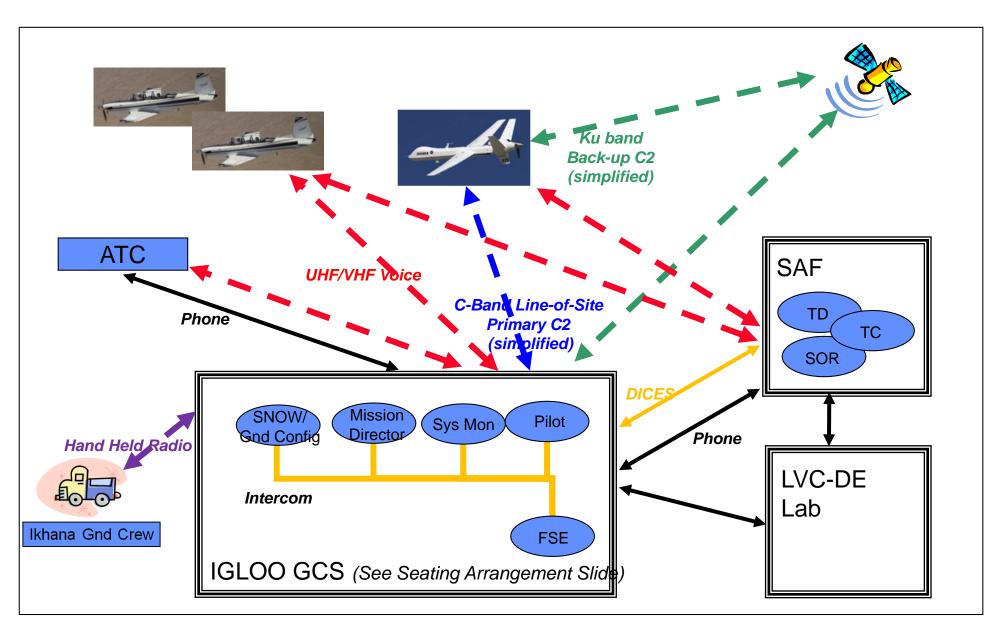
• Thales ADS-B receiver integrated

Quicklook 2

- MITLL product in development
- Operates using a stand alone computer
- May be installed prior to FT4
- Will not be used as a SA display for FT4



Ikhana UAS Communication – FT-4 Flights

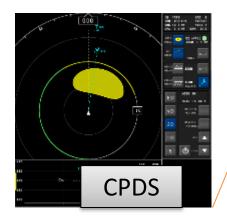




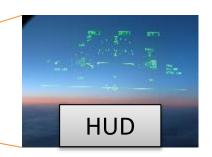
FT4 Displays Under Test











- DAA Display's are located to the left of PPO-1 Station
- HUD will display TCAS alerting. HDD's display TCAS WCA
- When specifically configured HUD will display DAA and TCAS alerting (Only 1 flight)
- Aural Alerts (Commands and Tone)
 - GCS will annunciate a TCAS alert tone
 - JADEM SUT will annunciate a TCAS alert commands



FT4 Displays Under Test

(Pilot/User centric)









DAA Alerts (VSCS)

Heading Bands "Turn Heading ###" DAA Alerts (VSCS / DAIDALUS Display)



Heading Bands

FT3 Capabilities



GCS Tone

DAA Alerting (VSCS)



Heading Bands

DAA Alerting (DAIDALUS Display)



Heading Bands

Descriptive

DAA Alerts

Lateral & Vertical



Vertical

FT4 Similarities



DAA Alerting

Vertical Bands WCR

Directive

TCAS Alerting

Vertical Commands

Aural Alerts

DAA Alerting

Descriptive

Vertical Bands



TCAS Alerting

Vertical Commands

Aural Alerts (May be muted) FT4 Differences

GCS Alerts Additional commands available

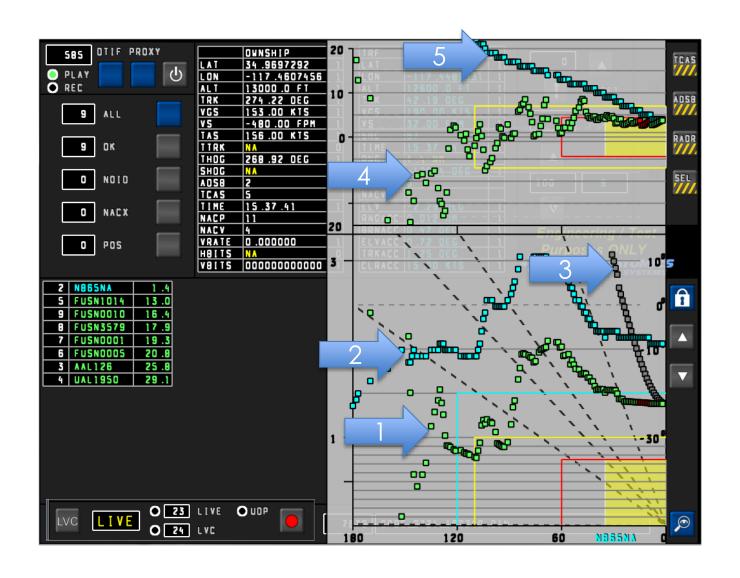
Descrip Directive

TCAS Alerting

DAA alerts displayed on HUD for 1 flight.

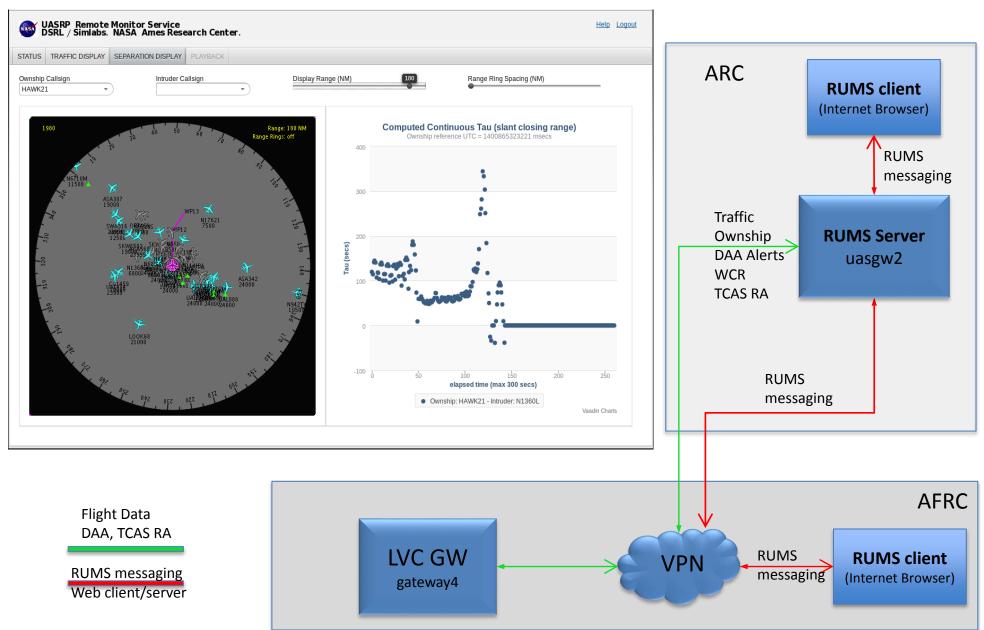


Winconverter with DCPA* plot enabled



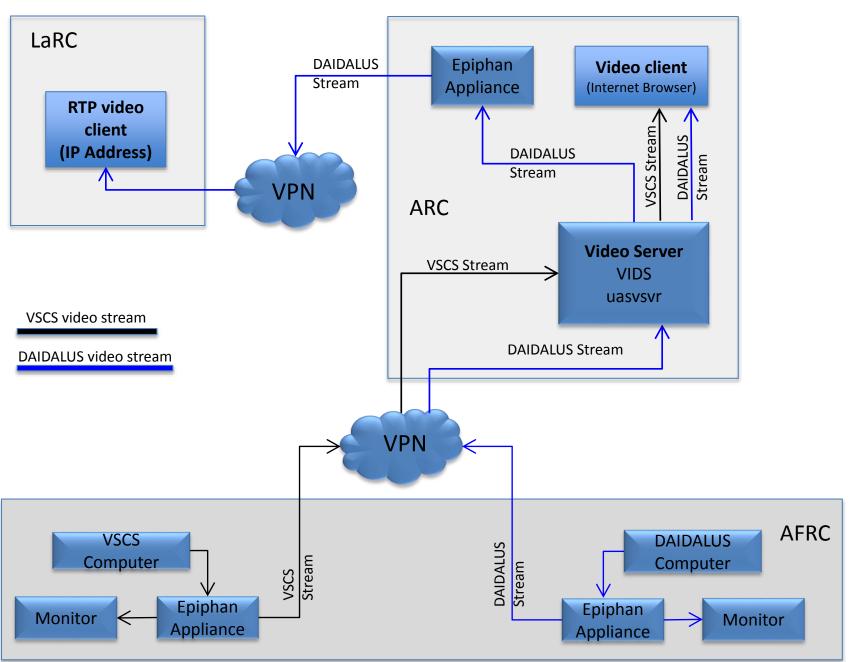


Configuration: RUMS



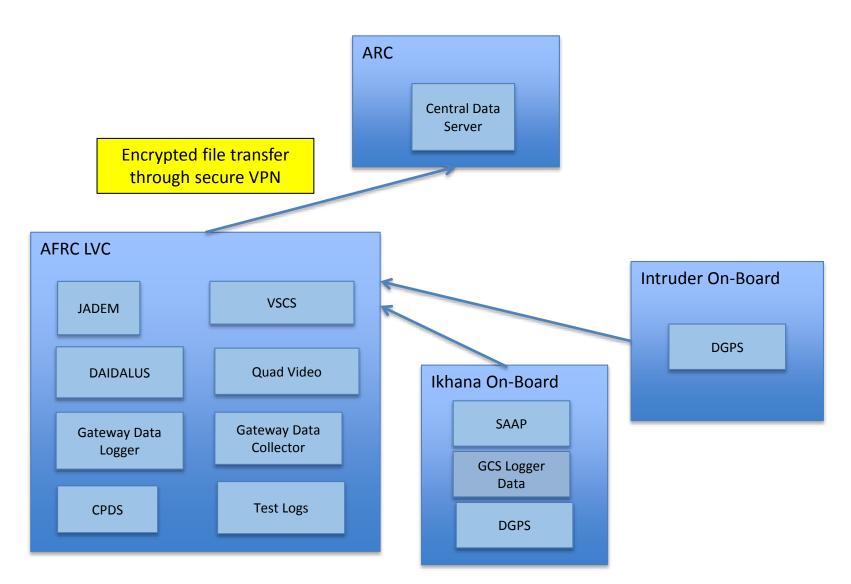


Configuration: Video Streaming





Data Collection Archiving



Reference: Flight Test Series 4 Data Management Plan, FT4 IT&E DMP-001



Flight Test 4 Flight Schedule



 Seeking approval from Tech Brief Board to fly FT4 Flight Test Series using current available integrated and tested version of LaRC provided DAIDALUS software.

•	FT4 Plan A (Baseline Version of DAIDALUS only)																									
									FT4	4 Plar	1 A (B	aselir	ne Ve	rsion	of DA	IDAL	US or	ıly)								
	April												М	ay			June									
	4 - 8		11 - 15		18 - 22		25 - 29		2 - 6		9 - 13		16 - 20		23 - 27		30 - 3		6 - 10		13 - 17		20 - 24		27 - 1	
Sys Check Flts																										
Env																										
Exp/Training																										
JADEM Flts																										
Radar Flts																										
CPDS Flts																										
TCAS Flts																										
HW Flts																										
SC-228 Flts																										
DAIDALUS Fits																	No Fly	(3-day k)								
DAIDALUS FILS																	VV	Ν,								

- LaRC working to update DAIDALUS software (final version) to be able to flight test with final version.
- Project approach:
 - Software needs to be delivered by 5/23 in order to meet FT4 completion milestone (L1) of 6/30/16.
 - 2-3 weeks of regression testing
 - Mini-tech updates
 - Complete DAIDALUS flights at the end of June

	FT4 Plan B (Final Version of DAIDALUS available by May 23)																									
	April												М	ay		June										
	4 - 8		11 - 15		18 -	22	25 - 29		2 - 6		9 - 13		16 - 20		23 - 27		30 - 3		6 - 10		13 - 17		20 - 24		27	- 1
Sys CheckFlts																										
Env																										
Exp/Training																										
JADEM Flts																										
Radar Flts																										
CPDS Flts																										
TCAS Flts																										
HW Flts																										0.5
SC-228 Flts																										35

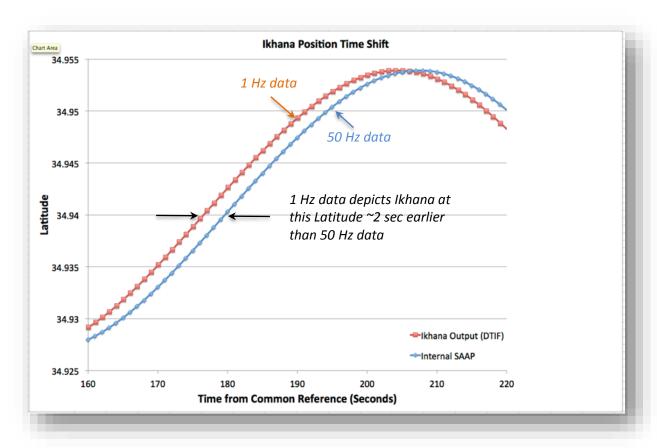


Backup Charts



Background Information

- During the FT3 Data Review, the IT&E co-PE from Ames identified a timing discrepancy in the ownship 1 Hz message containing position data (latitude, longitude) from Ikhana when compared to:
 - 10 Hz & 50 Hz ownship messages on Ikhana
 - Independent GPS data from Ikhana

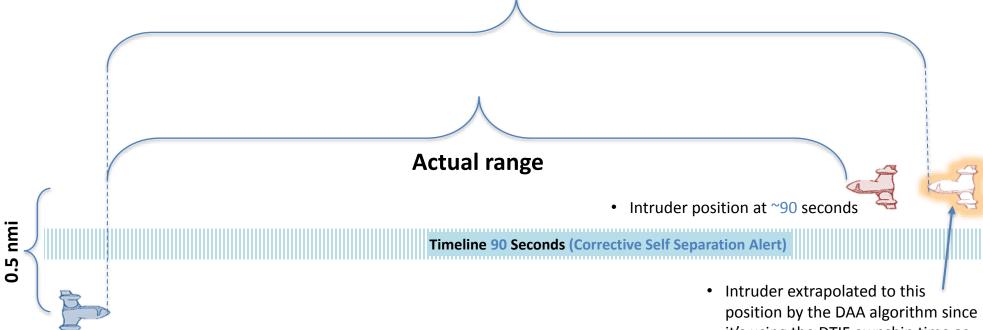


Graphs from Jim Murphy presentation 'FT3_data_analysis_v7.pptx'



Representation of Timing Discrepancy

Range perceived by DAA Algorithm

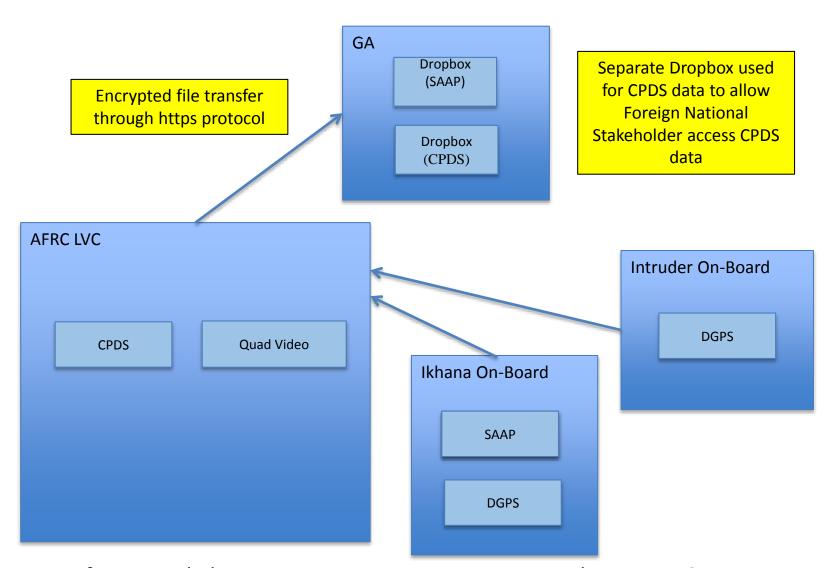


- Ownship position at T₀
- DTIF reports this position with time $T_0 = 2 \text{ seconds}$

it's using the DTIF ownship time as reference: ~(90 -~2 Seconds)



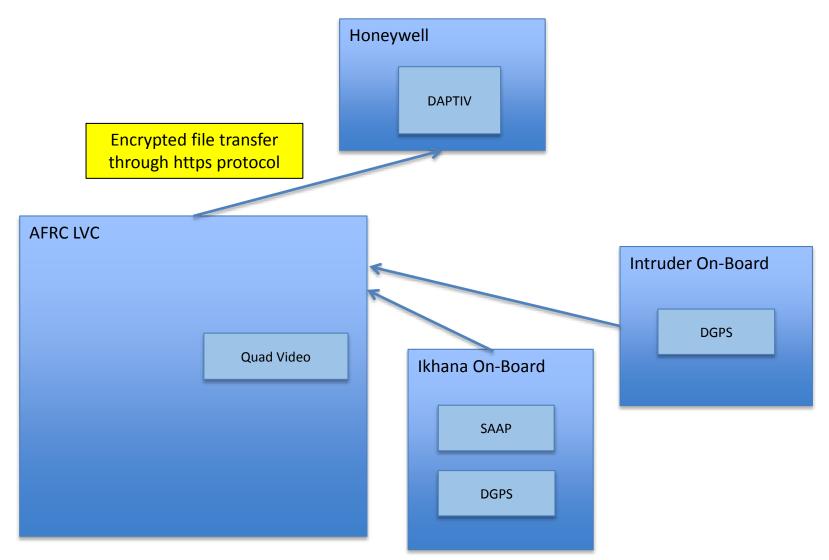
Data Collection Archiving



Reference: Flight Test Series 4 Data Management Plan, FT4 IT&E DMP-001



Data Collection Archiving



Reference: Flight Test Series 4 Data Management Plan, FT4 IT&E DMP-001